Parameters	Range	Installed system ¹ minimum accuracy (to recovered data)	Sampling inter- val (per second)	Resolution ³ read out	
Indicated airspeed	$V_{\rm m}$ in to $V_{\rm D}$ (KIAS) (minimum airspeed signal attainable with installed pilot-static system).	±5% or ±10 kts., whichever is greater.	1	1 kt.	
Altitude	-1,000 ft. to 20,000 ft. pressure altitude.	±100 to ±700 ft. (see Table 1, TSO C51-a).	1	25 to 150 ft.	
Magnetic heading Vertical acceleration	360° -3g to +6g	±5° ±0.2g in addition to ±0.3g maximum datum.	1	1°. 0.05g.	
Longitudinal acceleration	±1.0g	±1.5% max. range excluding datum error of ±5%.	2	0.03g.	
Pitch attitudeRoll attitude	100% of usable range ±60° or 100% of usable range, whichever is greater.	±2°±2°	1	0.8°. 0.8°.	
Altitude rate	±8,000 fpm	±10% Resolution 250 fpm below 12,000 ft. indicated.	1	250 fpm below 12,000.	
Engine Power, Each Engine		,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		,	
Main rotor speed	Maximum range Maximum range Maximum range	±5%	1	1% ² 1% ² 1% ²	
Primary (discrete) Secondary—if applicable (discrete).	High/low High/low		1. 1.		
Radio transmitter keying (discrete).	On/off		1.		
Autopilot engaged (discrete) SAS status—engaged (discrete). SAS fault status (discrete)	Engaged or disengaged Engaged/disengaged Fault/OK		1.		
Flight Controls					
Collective	Full range Full range Full range Full range	±3% ±3% ±3% ±3% ±3%	2 2 2 2	1% ² 1% ² 1% ² 1% ² 1% ²	

When data sources are aircraft instruments (except altimeters) of acceptable quality to fly the aircraft the recording system excluding these sensors (but including all other characteristics of the recording system) shall contribute no more than half of the values in this column.

2 Per cent of full range.

3 This column applies to aircraft manufactured after October 11, 1991.

 $[\mathrm{Doc.\ No.\ 25530,\ 53\ FR\ 26152,\ July\ 11,\ 1988;\ 53\ FR\ 30906,\ Aug.\ 16,\ 1988,\ as\ amended\ by\ Amdt.\ 13569,\ 62\ FR\ 38397,\ July\ 17,\ 1997]}$

APPENDIX D TO PART 135—AIRPLANE FLIGHT RECORDER SPECIFICATION

Parameters	Range	Accuracy sensor input to DFDR readout	Sampling inter- val (per second)	resolution 4 read	
Time (GMT or Frame Counter) (range 0 to 4095, sampled 1 per frame).	24 Hrs	±0.125% Per Hour	0.25 (1 per 4 seconds).	1 sec.	
Altitude	 1,000 ft to max certifi- cated altitude of air- craft. 	±100 to ±700 ft (See Table 1, TSO-C51a).	1	5' to 35' 1.	
Airspeed	50 KIAS to V_{so} , and V_{so} to 1.2 V_{D} .	±5%, ±3%	1	1kt	
Heading	360°	±2°	1	0.5°	
Normal Acceleration (Vertical)	-3g to +6g	±1% of max range excluding datum error of ±5%.	8	0.01g	
Pitch Attitude	±75°	±2°	1	0.5°	
Roll AttitudeRadio Transmitter Keying		±2°		0.5°.	
Thrust/Power on Each Engine	Full range forward	+2%	1 (per engine)	0.2%2.	

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Parameters	Range	Accuracy sensor input to DFDR readout	Sampling interval (per second)	resolution 4 read out
Trailing Edge Flap or Cockpit Control Selection.	Full range or each discrete position.	±3° or as pilot's indicator	0.5	0.5% ² .
Leading Edge Flap on or Cockpit Control Selection.	Full range or each discrete position.	±3° or as pilot's indicator	0.5	0.5% ² .
Thrust Reverser Position	Stowed, in transit, and reverse (discretion).		1 (per 4 sec- onds per en- gine).	
Ground Spoiler Position/ Speed Brake Selection.	Full range or each discrete position.	±2% unless higher accuracy uniquely required.	1	0.22 2.
Marker Beacon Passage Autopilot Engagement	Discrete		1	
Longitudinal Acceleration	±1g	±1.5% max range excluding datum error of ±5%.	4	0.01g.
Pilot Input And/or Surface Po- sition-Primary Controls (Pitch, Roll, Yaw) ³ .	Full range	±2° unless higher accuracy uniquely required.	1	0.2% 2.
Lateral Acceleration	±1g	±1.5% max range excluding datum error of ±5%.	4	0.01g.
Pitch Trim Position	Full range	±3% unless higher accuracy uniquely required.	1	0.3%².
Glideslope Deviation	±400 Microamps	±3%	1	0.3% 2.
Localizer DeviationAFCS Mode And Engagement Status.	±400 Microamps Discrete	±3%	1	0.3% 2.
Radio Altitude	-20 ft to 2,500 ft	±2 Ft or ±3% whichever is greater below 500 ft and ±5% above 500 ft.	1	1 ft + 5% ² above 500'.
Master Warning Main Gear Squat Switch Sta- tus.	Discrete		1	
Angle of Attack (if recorded directly).	As installed	As installed	2	0.3% 2.
Outside Air Temperature or Total Air Temperature.	-50° C to +90° c	±2° C	0.5	0.3° c
Hydraulics, Each System Low Pressure.	Discrete		0.5	or 0.5% ² .
Groundspeed	As installed	Most accurate systems installed (IMS equipped aircraft only).	1	0.2% 2.

If additional recording capacity is available, recording of the following parameters is recommended. The parameters are listed in order of significance:

		-		
Drift Angle	When available. As installed.	As installed	4	
Wind Speed and Direction	When available. As installed.	As installed	4	
Latitude and Longitude	When available. As installed.	As installed	4	
Brake pressure/Brake pedal position.	As installed	As installed	1	
Additional engine parameters:				
EPR	As installed	As installed	1 (per engine)	
N ¹	As installed	As installed	1 (per engine)	
N ²	As installed	As installed	1 (per engine)	
EGT	As installed	As installed	1 (per engine)	
Throttle Lever Position	As installed	As installed	1 (per engine)	
Fuel FlowTCAS:	As installed	As installed	1 (per engine)	
TA	As installed	As installed	1	
RA	As installed	As installed		
Sensitivity level (as se- lected by crew).	As installed	As installed	2	
GPWS (ground proximity warning system).	Discrete		1	
Landing gear or gear selector position.	Discrete		0.25 (1 per 4 seconds).	
DME 1 and 2 Distance	0-200 NM;	As installed	0.25	1mi.
Nav 1 and 2 Frequency Selection.	Full range	As installed	0.25.	

When altitude rate is recorded. Altitude rate must have sufficient resolution and sampling to permit the derivation of altitude to 5 feet.

2 Per cent of full range.

[Doc. No. 25530, 53 FR 26153, July 11, 1988; 53 FR 30906, Aug. 16, 1988]

APPENDIX E TO PART 135—HELICOPTER FLIGHT RECORDER SPECIFICATIONS

Parameters	Range	Accuracy sensor input to DFDR readout	Sampling interval (per second)	Resolution ² read out	
Time (GMT)	24 Hrs	±0.125% Per Hour	0.25 (1 per 4 seconds).		
Altitude	 1,000 ft to max certifi- cated altitude of air- craft. 	±100 to ±700 ft (See Table 1, TSO-C51a).		5' to 30'.	
Airspeed	As the installed meas- uring system.	±3%		1 kt	
Heading	360°	±2°	1	0.5°.	
Normal Acceleration (Vertical)	-3g to +6g	±1% of max range excluding datum error of ±5%.		0.01g	
Pitch Attitude	±75°	±2°	2	0.5°	
Roll Attitude	±180°	±2°	2	0.5°.	
Radio Transmitter Keying	On-Off (Discrete)		1	0.25 sec	
Power in Each Engine: Free Power Turbine Speed and Engine Torque.	0-130% (power Turbine Speed) Full range (Torque).	±2%	1 speed 1 torque (per engine).	0.2% ¹ to 0.4% ¹	
Main Rotor Speed	0–130%	±2%	2	0.3% 1	
Altitude Rate	±6.000 ft/min	As installed	2	0.2% 1	
	.,				
Pilot Input—Primary Controls (Collective, Longitudinal Cyclic, Lateral Cyclic, Pedal).	Full range	±3%	2	0.5% 1	
Flight Control Hydraulic Pressure Low.	Discrete, each circuit		1		
Flight Control Hydraulic Pressure Selector Switch Position, 1st and 2nd stage.	Discrete		1		
AFCS Mode and Engagement Status.	Discrete (5 bits nec- essary).		1		
Stability Augmentation System Engage.	Discrete		1		
SAS Fault Status	Discrete		0.25		
Main Gearbox Temperature Low.	As installed	As installed	0.25	0.5% 1	
Main Gearbox Temperature High.	As installed	As installed	0.5	0.5% 1	
Controllable Stabilator Position.	Full Range	±3%	2	0.4% 1.	
Longitudinal Acceleration	±1g	±1.5% max range excluding datum error of ±5%.	4	0.01g.	
Lateral Acceleration	±1g			0.01g.	
Master Warning	Discrete	datuiii 0i ±5%.	1		
Nav 1 and 2 Frequency Selection.	Full range	As installed	0.25		
Outside Air Temperature	-50° C to +90° C	±2° c	0.5	0.3° c	

[Doc. No. 25530, 53 FR 26154, July 11, 1988; 53 FR 30906, Aug. 16, 1988]

APPENDIX F TO PART 135—AIRPLANE FLIGHT RECORDER SPECIFICATION

The recorded values must meet the designated range, resolution, and accuracy requirements during dynamic and static conditions. All data recorded must be correlated in time to within one second. Accuracy (sensor Seconds per

Parameters	Range	input)	sampling interval	Resolution	Remarks
Time or Relative Time Counts Counts	24 Hrs, 0 to 4095.	±0.125% Per Hour.	4	1 sec	UTC time preferred when available. Counter increments each 4 seconds of system operation.

³For airplanes that can demonstrate the capability of deriving either the control input on control movement (one from the other) for all modes of operation and flight regimes, the "or" applies. For airplanes with non-mechanical control systems (fly-by-wire) the "and" applies. In airplanes with split surfaces, suitable combination of inputs is acceptable in lieu of recording each surface separately.

⁴This column applies to aircraft manufactured after October 11, 1991.

¹ Per cent of full range. ² This column applies to aircraft manufactured after October 11, 1991.